FTR - Flight Test Report Dieser Prüfbericht darf ohne schriftliche Zustimmung der EAPR nicht, auch nich

Manufacturer		Type testing No.	EAPR-GS-0796/18
	UP International Kreuzeckbahnstraße 7 D-82462 Garmisch-Partenkirchen	serial number	XD89-01-1-178-7417
Model	Meru - M	Location	Achensee und Bassano
Comment	glider was tested with foldinglines	Location	Monte Baldo, Gardasee



Rev. 2.3 - 26.11.2014 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing	15.01.2018	Minimum take off weight 100 kg			Maximum take off weight 115 kg			
Testpilot		Pascal Purin			Anselm Rauh		0	
Harness		EAPR schwer			EAPR			
Pilot's take off weigl	nt	100	kg		115	kg	aiv S	





Test-criteria	Minimum take off weight	Evaluation	Maximum take off weight	Evaluatio		
****	Millimum take on weight	Evaluation	Maximum take on weight	Evaluatio		
1. Inflation / take-off - 4.4.1	I.e		Le			
Rising behavior	Easy rising, some pilot correction is required	В	Easy rising, some pilot correction is required	В		
Special take off technique required	No	Α	No	Α		
2. Landing - 4.4.2						
Special landing technique required	No	Α	No	Α		
3. Speeds in straight flight - 4.4.3						
Trim speed more than 30km/h	Yes	А	Yes	Α		
Speed range using the controls larger than 10km/h	Yes	Α	Yes	Α		
Minimum speed	25 km/h to 30 km/h	В	25 km/h to 30 km/h	В		
4. Control movement - 4.4.4	<u> </u>			•		
Max. weight in flight up to 80kg		-		-		
Max. weight in flight 80 to 100kg		-		-		
Max. weight in flight greater than 100kg	Increasing 35cm - 50cm	D	Increasing 50cm - 65cm	С		
5. Pitch stability exiting accelerated flight - 4.4.5						
Dive forward angle on exit	Dive forward less than 30°	l A	Dive forward less than 30°	А		
Collapse occurs	No	Ä	No	A		
6. Pitch stability operating controls during accelera	ated flight - 4.4.6	,				
Collapse occurs	I No	l A	No	A		
7. Roll stability and damping - 4.4.7	110	, ,,	110	, ,,		
Oscillations	Reducing	A	Reducing	А		
	Treducing	A	Heddeling	A		
8. Stability in gentle spirals - 4.4.8			Lo.			
Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α		
9. Behaviour exiting a fully developed spiral dive				В		
Initial response of glider (first 180°)	No immediate reaction	В	No immediate reaction			
Tendency to return to straight flight	Turn remains constant	D	Turn remains constant	D		
Turn angle to recover normal flight	With pilot action	D	With pilot action	D		
10. Symmetric front collapse - 4.4.10						
Folding lines used	Yes	D	Yes	D		
Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α		
Recovery	Spontaneous in less than 3 sec 0° - 30° Keeping course	Α	Spontaneous in less than 3 sec	Α		
Dive forward angle on exit	E TOOPING COMME	Α	0° - 30° Keeping course	Α		
Cascade occurs	140	A	No No	A		
Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α		
Recovery	Spontaneous in 3 to 5 sec	В	Spontaneous in 3 to 5 sec	В		
Dive forward angle on exit	E DO DO Entering a turn or less than s		30° - 60° Entering a turn of less than 90°	В		
Cascade occurs	140	Α	No	Α		
Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α		
Recovery	3 sec	D	Recovery through pilot action in less than a furth 3 sec	U		
Dive forward angle on exit	g co co Entering a tam or co to rec	C	30° - 60° Entering a turn of 90° to 180°	C		
Cascade occurs	No No	Α	No	Α		
11. Exiting deep stall (parachutal stall) - 4.4.11						
Deep stall achieved	Yes		Yes			
Recovery	Recovery through pilot action in less than a fur 5 sec	D	Recovery through pilot action in less than a furth 5 sec	er D		
Dive forward angle on exit	30° - 60°	В	0° - 30°			
Change of course	Changing course less than 45°	Α	Changing course 45° or more	С		
Cascade occurs	No	Α	No	Α		

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12. High angle of attack recovery - 4.4.12									
Recovery		Recovery through pilot action in less than a further			D	Recovery throu 5 sec	D		
Cascade occurs		5 sec No			Α	No Sec	Α		
13. Recovery from a developed full stall - 4.4.13									
Dive forward angle on exit Collapse		30° - 60° No collapse			B A	30° - 60° No collapse			B A
Cascade occurs (other than collapse)		No			Α	No			Α
Rocking backward Line tension		Less than 45° Most lines tight			A A	Less than 45° Most lines tight			A
14. Asymmetric collapse (trim speed) - 4.4.14	West miss tight				moot moo tigrit				
Folding lines used		Yes	Yes			Yes			D
Change of course until re-inflation	98	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re-	-inflation		Α	Spontaneous re	-inflation		Α
Total change of course	sbe % cc	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs	trin ax 51	No			Α	No	Α		
Twist occurs Cascade occurs	Ε	No No			A	No No			A
Change of course until re-inflation		90° - 180°	Dive or roll angle	45° - 60°	С	90° - 180°	Dive or roll angle	45° - 60°	С
Re-inflation behavior	trim speed, max 75% collapse	Spontaneous re-	-inflation		A	Spontaneous re	-inflation		Α
Total change of course	trim speed,	Less than 360°			Α	Less than 360°		Α	
Collapse on the opposite side occurs Twist occurs	trir nax 7	No No			A	No No			A
Cascade occurs	L	No No			A	No	A		
Change of course until re-inflation		< 90°	Dive or roll angle	15° - 45°	А	< 90°	Dive or roll angle	15° - 45°	Α
	d, apse		1	10 - 40				10 - 40	
Re-inflation behavior	accelerated, max 50% collapse	Spontaneous re-	-inflation		Α	Spontaneous re	e-inflation		Α
Total change of course Collapse on the opposite side occurs	ccele 50%	Less than 360° No			A	Less than 360° No			A
Twist occurs	max	No			A A	No			A
Cascade occurs		No			Α	No	ı		Α
Change of course until re-inflation	esd.	90° - 180°	Dive or roll angle	60° - 90°	D	90° - 180°	Dive or roll angle	60° - 90°	D
Re-inflation behavior	accelerated, max 75% collapse	Spontaneous re-	-inflation		Α	Spontaneous re	e-inflation		Α
Total change of course Collapse on the opposite side occurs	accel x 75%	Less than 360°			A	Less than 360° No			A
Twist occurs	max	No			Α	No	Α		
Cascade occurs 15. Directional control with a maintained asymmetry.	matric cal	No			Α	No			Α
Able to keep course straight	netric coi	Yes			Α	Yes			Α
180° turn away from the collapsed side possible in	10 sec	Yes			Α	Yes			A
Amount of control range between turn and stall or spin		More than 50% of the symmetric control travel			Α	More than 50%	A		
-	Wide than 50 %	or the symmetric of	onitioi tiavei	Α	More than 50 %	^			
16. Trim speed spin tendency - 4.4.16 Spin occurs	No			Α	No	Α			
17. Low speed spin tendency - 4.4.17									
Spin occurs		No			Α	No			Α
18. Recovery from a developed spin - 4.4.18		ı				ı			
Spin rotation angle after release		Stops spinning in 90° to 180°			С	Stops spinning i	С		
Cascade occurs 19. B-line-stall - 4.4.19	No			Α	No	Α			
19. B-line-stall - 4.4.19 Change of course before release		1			NA				NA
Behaviour before release					NA			NA	
Recovery					NA		NA		
Dive forward angle on exit					NA		NA		
Cascade occurs					NA				NA
20. Big ears - 4.4.20									
Entry procedure		Standard technique			A	Special device	A		
Behaviour during big ears		Stable flight Recovery through pilot action in less than a further			A	Stable flight	A		
Recovery		3 sec			В	Spontaneous in	В		
Dive forward angle on exit 0° - 30° 21. Big Ears in accelerated flight - 4.4.21				Α	0° bis 30°	Α			
Entry procedure		Standard technique			Α	Special device	Α		
Behaviour during big ears		Stable flight			Α	Stable flight	- L - 20-1		Α
Recovery		Spontaneous in 3 to 5 sec			Α	Recovery throu 3 sec	gh pilot action in les	ss than a further	В
Dive forward angle on exit		0° - 30°			Α	0° bis 30°			Α
Behaviour immediately after releasing the accelara maintaining big ears	Stable flight			Α	Stable flight	Α			
23. Alternative means of directional control - 4.4.22									
180° turn achievable in 20 sec Yes				Α	Yes			Α	
Stall or spin occurs No A No A								Α	
23. Any other flight procedure and/or configura	ation desc	cribed in the user	's manual - 4.4.2	3	N/A				NIA
Procedure works as descibed Procedure suitable for novice pilots				NA NA				NA NA	
Cascade occurs				NA					
24. Remarks of testpilot:									
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