FTR - Flight Test Report Dieser Prüfbericht darf ohne schriftliche Zustimmung der EAPR nicht, auch nich

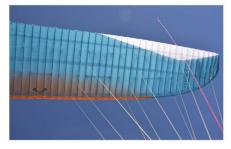
Manufacturer	\(\sqrt{\text{DVANCE}} \)	Type testing No. EAPR-GS-032			
	Advance Thun AG Uttingenstr. 87 CH-3600 Thun	serial number	61440		
Model	IOTA 28	Logotion	Schruns		
		Location	Krippenstein, Hallstätter See		



Rev. 2.2 - 09.10.2014 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing	14.10.2014	Minimum take 90 kg	eight	Maximum take off weight 115 kg			
Testpilot		Hannes Tschofen			Anselm Rauh		
Harness		EAPR		150	EAPR schwer		de la
Pilot's take off weigh	nt	90	kg		115	kg	A VI





Test-criteria		Minimum take off weight	Evaluation	Maximum take off weight	Evaluation	
1. Inflation / take-off - 4.4.1						
Rising behavior		no pilot correction required	А	no pilot correction required	А	
Special take off technique required		No	Α	No	Α	
2. Landing - 4.4.2						
Special landing technique required		I No	A	No	A	
3. Speeds in straight flight - 4.4.3		140	Α	140		
Trim speed more than 30km/h		Yes	l A	Yes	l A	
Speed range using the controls larger than 10	km/h	Yes	Α	Yes	Α	
Minimum speed		Less than 25 km/h	А	Less than 25 km/h		
4. Control movement - 4.4.4						
Max. weight in flight up to 80kg			-		-	
Max. weight in flight 80 to 100kg					-	
Max. weight in flight greater than 100kg		Increasing >65 cm	А	Increasing >65 cm	А	
5. Pitch stability exiting accelerated flight	- 4.4.5	<u> </u>				
Dive forward angle on exit		Dive forward less than 30°	А	Dive forward less than 30°	А	
Collapse occurs		No	Α	No	Α	
6. Pitch stability operating controls during	accelerated	flight - 4.4.6				
Collapse occurs		No	Α	No	Α	
7. Roll stability and damping - 4.4.7						
Oscillations		Reducing	А	Reducing	A	
8. Stability in gentle spirals - 4.4.8			•			
Tendency to return to straight flight		Spontaneous exit	A	Spontaneous exit	A	
9. Behaviour exiting a fully developed spir	al dive - 4.4.	9				
Initial response of glider (first 180°)		No immediate reaction	В	No immediate reaction	В	
Tendency to return to straight flight			A		A	
Turn angle to recover normal flight	Turn angle to recover normal flight		Α		В	
10. Symmetric front collapse - 4.4.10						
Folding lines used		No		No		
Entry	30%	Rocking back less than 45°	Α	Rocking back less than 45°	Α	
Recovery		Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	А	
Dive forward angle on exit	ruim speed	0° - 30° Keeping course	Α	0° - 30° Keeping course	Α	
Cascade occurs	E	No	A	No	A	
Entry	% 20%	Rocking back less than 45°	A	Rocking back less than 45°	A	
Recovery	× peeds	Spontaneous in 3 to 5 sec	В	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	ds min	0° - 30° Keeping course	Α	0° - 30° Keeping course	A	
Cascade occurs		No.	A	No	A	
Entry	20%	Rocking back less than 45°	A	Rocking back less than 45°	A	
Recovery	ocelerated > 50%	Spontaneous in 3 to 5 sec	В	Spontaneous in less than 3 sec	А	
Dive forward angle on exit	ocele	0° - 30° Keeping course	A	30° - 60° Keeping course	В	
Cascade occurs		No	Α	No	A	
11. Exiting deep stall (parachutal stall) - 4.	4.11					
Deep stall achieved		Yes		Yes		
Recovery		Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec		
Dive forward angle on exit		0° - 30°	A	0° - 30°	А	
Change of course		Changing course less than 45°	Α	Changing course less than 45°	Α	

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12. High angle of attack recovery - 4.4.12									
Recovery				А	Spontaneous in less than 3 sec				
Cascade occurs	No			A	No			A	
13. Recovery from a developed full stall - 4.4.1					140				
Dive forward angle on exit	30° - 60°			В	30° - 60°			В	
Collapse		No collapse			A	No collapse			A
Cascade occurs (other than collapse) Rocking backward		No Less than 45°			A	No Less than 45°			A
Line tension		Most lines tight			A	Most lines tight			A
14. Asymmetric collapse (trim speed) - 4.4.14									
Folding lines used	1	No	1	ı		No	1	I	
Change of course until re-inflation	trim speed, max 50% collapse	< 90°	Dive or roll angle	0° - 15°	Α	< 90°	Dive or roll angle	0° - 15°	Α
Re-inflation behavior		Spontaneous re-	-inflation		А	Spontaneous re	-inflation		Α
Total change of course	rim speed c 50% colls	Less than 360° No No No			A	Less than 360°			Α
Collapse on the opposite side occurs	ax 5(Α	No			А
Twist occurs Cascade occurs	Ë				A	No No			A
		90° - 180°		450 450				450 450	
Change of course until re-inflation	pse	90° - 180°	Dive or roll angle	15° - 45°	В	90° - 180°	Dive or roll angle	15° - 45°	В
Re-inflation behavior	trim speed, max 75% collapse	Spontaneous re-	-inflation		Α	Spontaneous re	-inflation		Α
Total change of course	n sp	Less than 360°			А	Less than 360°			А
Collapse on the opposite side occurs	trim lax 75°	No			A	No No No			A
Twist occurs Cascade occurs	Ε	No No			A				A
		I							
Change of course until re-inflation	Se	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	accelerated, max 50% collapse	Spontaneous re-	-inflation		А	Spontaneous re	-inflation		Α
Total change of course	accelerated, x 50% collap	Less than 360°			A	Less than 360°			Α
Collapse on the opposite side occurs	acc ax 50	No			A	No No			A
Twist occurs Cascade occurs	Ë	No			A				A
		No	1		A	No			A
Change of course until re-inflation	se	90° - 180°	Dive or roll angle	15° - 45°	В	90° - 180°	Dive or roll angle	15° - 45°	В
Re-inflation behavior	accelerated, max 75% collapse	Spontaneous re-	-inflation		Α	Spontaneous re	-inflation		Α
Total change of course	accelerated x 75% colla _l	Less than 360° No No			А	Less than 360° No			A
Collapse on the opposite side occurs	acc lax 7				A				Α
Twist occurs Cascade occurs	Ε				A	No No			A
15. Directional control with a maintained asymm	metric col					110			
Able to keep course straight		Yes			А	Yes			Α
180° turn away from the collapsed side possible in	10 sec	Yes			А	Yes			Α
Amount of control range between turn and stall or	spin	More than 50% (of the symmetric	control travel	А	More than 50%	of the symmetric	control travel	Α
16. Trim speed spin tendency - 4.4.16									
Spin occurs		No			А	No			Α
17. Low speed spin tendency - 4.4.17 Spin occurs		No			А	No	Α		
18. Recovery from a developed spin - 4.4.18		NO			A	140			
Spin rotation angle after release		Stops spinning in less than 90°			Α	Stops spinning in less than 90°			Α
Cascade occurs	No			A	A No			Α	
19. B-line-stall - 4.4.19 Change of course before release		Changing course	loss than 45°		А	Changing course	a lose than 45°		Α
		Changing course less than 45° Remains stable with straight span				Remains stable with straight span			
Behaviour before release		nemains stable	witti straight span	1	А	nemains stable	witti straignt span		Α
Recovery		Spontaneous in	less than 3 sec		Α	Spontaneous in	aneous in less than 3 sec		
Dive forward angle on exit		0° - 30°			А	0° - 30°			
Cascade occurs No				А	No			Α	
20. Big ears - 4.4.20									
Entry procedure	Special device re	equired		Α	Special device required			Α	
Behaviour during big ears	Behaviour during big ears		Stable flight			Stable flight			Α
Recovery		Spontaneous in	3 to 5 sec		В	Spontaneous in less than 3 sec			Α
Dive forward angle on exit		0° - 30°			Α	0° bis 30°			Α
21. Big Ears in accelerated flight - 4.4.21									
Entry procedure	Special device re	equired		А	Special device r	equired		Α	
Behaviour during big ears		Stable flight		A	Stable flight			A	
Recovery		Spontaneous in 3 to 5 sec		A	Recovery through pilot action in less than a further			В	
·		·			3 sec				
Dive forward angle on exit Behaviour immediately after releasing the accelarator while		0° - 30°		Α	0° bis 30°			A	
maintaining big ears Stable flight			А	Stable flight			Α		
23. Alternative means of directional control - 4.4.22									
180° turn achievable in 20 sec Yes				Α	Yes			Α	
Stall or spin occurs No					А	No		Α	
23. Any other flight procedure and/or configura	ation desc	cribed in the user	's manual - 4.4.	.23					
Procedure works as descibed					NA				NA
Procedure suitable for novice pilots Cascade occurs					NA NA				NA NA
24. Remarks of testpilot:								- 31	
		L				L			

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